## GBCS SCHEWE

USN						18AE34

## Third Semester B.E. Degree Examination, Jan./Feb. 2023

**Elements of Aeronautics** Time: 3 hrs. Max. Marks: 100 Note: Answer any FIVE full questions, choosing ONE full question from each module. Draw a neat sketch of a typical aircraft and mention the propose of each part. (10 Marks) What are high lift devices? Explain with the help of sketches. (10 Marks) Draw a neat sketch of a typical wing and mention the purpose of each part. 2 (12 Marks) b. Explain the desirable properties for aircraft. (08 Marks) Module-2 What are the forces acting on aircraft? Explain drag in detail. (10 Marks) Determine the standard atmospheric values of density, temperature at a geopotential altitude of 15km. (10 Marks) Define the following terms: Aerodynamic center, center of pressure, aspect ratio, Mach b. The pressure at a point on wing of an airplane flying at a velocity of 70m/s in the conditions associated with a standard altitude of 2km is 7.58 × 10<sup>4</sup>N/m<sup>2</sup>. Calculate the pressure coefficient at that point on the wing. Sea level values of density, temperature and pressure are respectively, 1.225kg/m<sup>3</sup>, 288.16 K and 1 bar. Take  $\alpha = -6.5$ k/KM. (10 Marks) Module-3 Explain the aircraft classification based on power-plant location. (10 Marks) Compare the merits and demerits of turbojet and turbofan engines. (10 Marks) Describe with neat sketch the working of turboprop engine. (10 Marks) b. Explain the Ramjet with neat sketch. (10 Marks) Module-4 Explain the lateral stability with neat sketch. 7 (10 Marks) Explain the purpose of Tabs in aircraft. (10 Marks)

## OR

- Explain the turn performance and derive the relations for the radius of turn and turn rate. 8
  - (10 Marks) The maximum L/D of an airplane is 15. Calculate the minimum glide angle and maximum range measure along the ground by airplane in a power off glide at altitude of 4000m. Also calculate equilibrium glide velocity. Given density at 4km altitude is 0.8193kg/m<sup>3</sup>, gross weight of airplane is 1360kg, wing surface area is  $17\text{m}^2$  and  $C_L = 0.65$ . (10 Marks)

Module-5

Describe typical hydraulic system with neat sketch. (10 Marks)

Describe a typical pneumatic system of an aircraft with neat sketch.

(10 Marks)

OR

Explain FBW with a neat sketch. 10

(10 Marks)

Explain the inertial navigation system employed in aircraft.

(10 Marks)