

Important Note : 1. On completing your answers, compulsorily draw diagonal cross lines on the remaining blank pages. 2. Any revealing of identification, appeal to evaluator and /or equations written eq. 47+8 = 50. will here

A pitot-static tube is used to measure the velocity of water in a pipe. The stagnation pressure c. head is 6 m and static pressure head is 5 m. Calculate the velocity of flow assuming the (04 Marks) co-efficient of tube equal to 0.98

OR

- Define Reynold's number. What is its significance? List the characteristics of laminar flow. 6 a. (08 Marks)
 - A fluid of viscosity 0.7 NS/m² and specific gravity 1.3 is flowing through a circular pipe of b. diameter 100 mm. The maximum shear stress at the pipe wall is given as 196.2 N/m². Find, (i) The pressure gradient, (ii) The average velocity (iii) Reynold's number of the flow. (08 Marks)
 - A rough pipe is of diameter 8.0 cm. The velocity at a point 3.0 cm from wall is 30% more C. than the velocity at a point 1 cm from pipe wall. Determine the average height of the (04 Marks) roughness.

Module-4

- Define Lift force and drag force. Also derive their expressions. (08 Marks) 7 a.
 - (ii) Turbulent boundary layer Define: (i) Laminar boundary layer. b. (iv) Boundary layer thickness (08 Marks) (iii) Laminar sub-layer
 - A flat plate 1.5 m×1.5 m moves at 50 km/hour in stationary air of density 1.15 kg/m³. If the C. co-efficient of drag and lift are 0.15 and 0.75 respectively, determine :
 - The lift force. (i)

resisting force.

- The drag force. (ii)
- The resultant force. (iii)
- The power required to keep the plate in motion. (iv)

OR

- Explain dimensional homogeneity with examples. 8 a.
 - Define similitude and explain the following : b.
 - (i) Geometric similarity (ii) Kinematic similarity (iii) Dynamic similarity (08 Marks) The resisting force (R) of a supersonic plane during flight can be considered as dependent C. upon the length of the aircraft (l), Velocity (V), Air Viscosity (μ), Air density (ρ) and Bulk modulus of air (K). Express the functional relationship between these variables and the

Module-5

- Define : (i) Mach number (ii) Mach cone (iii) Zone of action (iv) Super-sonic flow 9 a (04 Marks)
 - b. Define stagnation properties. Obtain an expression for stagnation pressure of a compressible (08 Marks) fluid in terms of Mach number and pressure.
 - A projectile travels in air of pressure 10.1043 N/cm² at 10°C at a speed of 1500 km/hour. C. Find the Mach number and Mach angle. Take K = 1.4 and $R = 287 \text{ J/kg}^{\circ} \text{K}$. (08 Marks)

OR

(04 Marks) What is CFD? Explain the necessity of CFD. 10 a. List the advantages, disadvantages and applications of CFD. (08 Marks) b. Write short note on "Engineering philosophy behind CFD. (08 Marks) C.

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(04 Marks)

(08 Marks)

(04 Marks)